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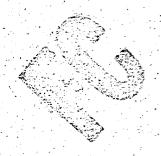
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NEW INFRARED FLARE
AND MIGH-ALTITUDE
IGNITER COMPOSITIONS (U)

CHARLES A. KNAPP

JULY 1959





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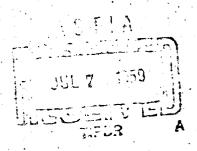
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NEW IMPRARED PLANE AND MIGH-ALTITUDE IGNITER COMPOSITIONS (U)

by

Charles A. Kango

July 1950

Poitura Recorreb and Englocertag Laboresarios Plantinty Areasad Dover, N. J.

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(S) OBJECT

To conduct research on and to develop infrared flure decoy systems with energy output rich in the 1.2 - 2.8 m spectral region for the protection of jet aircraft egainst intrared-seeking missiles. (This work is being conducted under Air Force contract (33-616) 59-45.)

(S) SUMMARY

(S) Breakthroughs have been ackieved with both infrared flare compositions and high-altitude igniter compositions. Preliminary tests of a new flare composition (SI-119) show that its burning rate and high ignitability are unaffected by increaving altitude up to 100,000 feet (6 mm Hg) and that its peak infrared energy output (watts) and efficiency (joules/g and cc) increase with increasing altitude. The new composition, which consists of molybdenum trioxide/chromic oxide/zirconium, shows considerable promise for use in decoy systems for jet aircraft. Both the currently used tellon/ magnesium/nitroceilulose compositions and the older, less efficient rodium nitrate/magnesium/Laminac compositions are adversely affected by increasing altitude: peak energy output and efficiency decrease severely, burning race decreases greatly, and the flares become very difficult to ignite. At high altitude the SI-119 composition is far superior to teflon compositions, especially during the first 12 seconds of burning.

(C) New high-altitude igniter compositions containing molybdenum trioxide/ chromic oxide/zircuaium/actrocellulos-(SI-122 and 131) have also been formulated which ignite tellon compositions at altitudes up to 100,000 feet (highest altitude simulated). These new igniter compositions should perform as well or better with sodium nitrate compositions. They are superior to the 90/10/5 barium chromate/boron/nitrocellulose composition which, unless partially confined, is unreliable above 40,000 feet altitude and is more sensitive to frictional forces.

(C) RECOMMENDATIONS

In view of the encouraging results with respect to burning characteristics and energy output obtained with new infrared flare compositions containing molybdenum trioxide/chromic oxide/zirconium furtheresearch and development should be conducted. Fith higher energy oxidizing agents this type of composition may equal or exceed the total efficiency of the teflon composition at lower alricudes as well as at 100,000 feet. It is also expected that the advantages of burning rates unaffected by decreasing pressure plus excellent ignitability could be maintained.

It is further recommended that more extensive tests be conducted with the new high-altitude igniter compositions (MoO₂/Cr₂O₃/Zr/NC) to determine optimum composition and reliability. The use of coarser zirconium should be investigated to further reduce sensitivity to friction and impact.

The use of conductive igniter

Although research and development of projection systems (for use with jet aircraft) are also continuing, only the most outstanding flace and igniter composition developments are discussed in this report.

(C) RESULTS AND DISCUSSION

Development of Flare Composition

- 4. Composition formulations with heats of reaction and heats of combustion data are listed in Table 1 (p. 7). Since the calorimetry teras were conducted under coatined coaditions, these results probabiy represent upper energy limits. At high altitudes where vacuum conditions exist, combustion efficiency and reaction energy are reduced due to lower flame temperatures and much slower burning rates. This was shown in a previous repen (Ref 2) where a composition consisting of 54% magersiam, 50% tellon, and 16% Kel F gave an average heat of reaction in one atmosphere of helium of 1521 calories/gram and at 5 millimeters of helium of 1212 culories/gram.
- 5. Further evidence of the deleterious effects of reduced pressure on burning characteristics can be seen in Table 2 (p 8) and Figure 1 (p 12). The efficiency of sodium nitrate/magnesium/Laminac and teflon/magnesium/aitrocellulose compositions in the form of pressed flares decreases sharply with increasing altitude (decreasing pressure). The peak infrared energy output of the teflon

- level to 60,000 feet and 95% from sea level to 100,000 feet. In addition, i. s burning rate decreases approximately 2 and 7 times, respectively. Nevertheless, the teflon/magnesium/nitrocellulose system is far superior to the exclum nitrate system. This may be due to the large quantity of incandescent carbon (high emissivity) produced by the resection of the teflon composition.
- 6. A radical departure from the above trends occurred with a composition consisting of MoO₃/C₇,O₃/Zr (SI-119). This composition became significantly more efficient with increasing altitude. Peak energy output and efficiency increased ap to 60,000 feet and tapered off slightly rs 100,000 feet. The burning rate of the SI-119 composition remained essentially unchanged at the higher altitudes. These effects can be seen in Table 2 (p 8) and Figure 2 (p13). It is important to note that whereas at sea level composition SI-119 was much less efficience and produced a much lower peak energy output (in watts) than the teflon system, at 60,000 feet the peak energy of SI-119 was about 4 times higher than that of the teflon system (although efficiency was still lower). At 100,000 feet (simulated altitude) the peak energy level of SI-119 was about 20 times greater than that of the teflon system, and its efficiency (in joules/cc) was approximately the same (over the entire burning period). If only the first 12 seconds of burning are considered it can be seen from Table 3 (p 9) that composition

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Ni 119 is far superior to the tellon system in peak energy output at 60,000 and 100,000 feet, and efficiency at 100,000 feet. All flaces in the above study were end burning and had approximately the same burning surface area.

- 7. Table 1 (p. 7) shows Composition SI-119 to have a relatively low heat of reaction and combustion. It is anticipated that its reaction energy can be greatly increased by incorporating chemicals producing higher energy. One approach is the use of perchlorates as oxidizing agents. For example, KClO. with zirconium produces 1610 calories/ gram and 6150 calories/cc (based on calculated true density). Other perchloraces can give even higher energies (Ref 3). Another approach is the use of superoxides of the first two periodic groups. For example, the calculated heat of reaction for KO, and zirconium is 1170 catories/gram. The use of higher oxygen complex compounds of molybdenum (MaMoOa) with the alkali metals should also be investigated. In addition, the use of excess zirconium may also be of value. Composition SI-119 is approximately stoichiometric. assuming ZrO, Mo, and Cr as products. Still another approach to the optimum composition is a mixture of the zirconium (SI-119) and the teflon compositions.
- 8. Another important advantage of Composition SI-119 is that it is highly ignitable. It is also used as an igniter composition and can be initiated directly

- by an M1A1-type aquib (containing 90/10 barium chromate/boron) up to 100,000 feet. It is believed that aquibs or primers could be eliminated if desired, since this new composition can probably be ignited by hot wires (squib principles Simplifying the system by eliminating components might logically lead to increased reliability. To slow down the relatively fast burning rate of SI-119 and to reduce its sensitivity to friction courses zirconium (Ref 4) and binders could be used.
- 9. Composition SI-119, which contains no binder, was pressed into a paper flare case for testing. If necessary, it can be converted into a caseless flare by means of binders in the same manner as the sodium nitrate/magnesium and teflon/magnesium compositions. To meet requirements, binders can also be used to adjust the burning rate. It is anticipated, however, that even with the incorporation of binders the burning rate of composition SI-119 will remain relatively unaffected by changes of altitude over the range of sea level to 100,000 feet.

Development of Igniter Composition

10. Two new igniter compositions containing NoO₃/Cr₂O₃/Zr/nitrocellulose (SI-122 and SI-131) have been developed for use with teflon/magnesium/nitrocellulose flare formulations. Although these igniter compositions were not tested with sodium nitrate/magnesium/haminac flare compositions the results of Reference 4 indicate that they should

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perform even better with flores of this type. Static igaition test results with flare compositions are given in Table 4 (p 10). Ignition of the flare compositions was accomplished over the simulated altitude range of sea level to 100,000 feet with completely unconfined systems. At 100,000 feet, however, ignition of the flaces by the igniter was erratic. Preliminary tests with these igniters indicate that at 100,000 feet the type of surface of the flare may be critical. Flares with porous surfaces ignited consistently while those with hard glazed surfaces did nor. It should be pointed out that when the standard Rita igniter composition, SI-56 (berium chromate/boron/ nitrocellulose), is completely unconfined, it performs erratically between 40,000 and 60,000 feet. Above 60,000 feet it fails completely to ignite the flare composition.

11. Data on the sensitivity of the igniter compositions to impact and frictional forces is given in Table 5 (p. 11). It can be seen that the SI-122 igniter composition (MoO₃/Cr₂O₄/Zr/nitrocellulose) is less sensitive than SI-56 (BaCrO₄/B/nitrocellulose). SI-56 has an impact value of 3 inches and SI-122 an impact value of 12 inches. SI-56 is sensitive to both steel and fiber (friction pendulum test) whereas SI-122 is sensitive to steel only.

(C) EXPERMIENTAL PROCEDURE

12 (U) Compositions containing binder were prepared in a Luncaster blender.

A small amount of accione (10 cc/100) g of mix) was used to insure proper blanding. Blending time was approximately 20 minutes.

13. (U) Binderless compositions were blended in an Abbe ball mill for 1/2 hour. Rubber stoppers were used to insure proper blending.

14. (C) SI-122 and 131 were blended by stirring with a wooden rod in a mortar. The nitrocellulose was used as a 10% solution in acetone. Excess sectone (up to 30 cc/100 g mix) was used for blending. While still wet with acetone these compositions were applied to the flare sides with a paint treah.

15. (C) Caseless flores were pressed in one increment at 7,000 psi. Cased flores were pressed in 1-inch increments at 15,000 psi.

16. (C) Materials

NaNO, USP, DR, 23 micron	Davies Nitrate Co
MoO ₃ , 12.5 micron	Pinher Scientific
	Co., Lot 762937
Cz ₂ O ₃ , less than 1 micros	Fisher Scientific
	Co., Let 516340
Teflon, represented. 25.	Device Nitrets Co
52, and 92 micros	
Kol-F, pulvetized, 28 micros	Kellogg Co.,
	Lot 4980-14
Mg, esemized, 23 micron.	Ruffert Chemical
granulacion 200/325	Ca.
Mg, sectioned, 100 micron.	Rufter Chemeral
granulation 50/100	Co.

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Es, 2.3 micron	Force Macrai
	Co., Grade Z,
	Let 70262
Suroceliulose, 12.6%	ŧ
aurogea, lintera	
Laminac Resin Nov. 4116	American Cy-
and 4134, commercial grade	enemid Corp.

and 4134, commercial grade

17. (C) All flares were tested in an upright position (Fig 4, p 15) in the Picatinny high-altitude test chamber (Fig. 3, p.14). Energy measurements were taken from a horizontal side-on position, using lead sulfide detection equipment.

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TABLE 1

Composition Formulas and Reaction Energies

Composition, phw	Perticle Size, microns	FY-719	FY-498	FY-893	FW-155	FW-154	FW-146	F W- 168	šI-119	\$1-122	\$! 1
NaNO,	23	47.6	33.5	60	-	-		_	••	_	_
MoO ₃	12	-	-	~	_	**	_	-	31.5	31.3	25
$Cr_{2}O_{1}$	0.5	~	-	-	-		-	_	20	20	16
Teilon	23		-	-	46 ^a	30 ª	46ª	46	_	_	
Kel-F	28	-	-	_	_	16	-	-	-	-	-
Mg	23	47.6	61.7 ^b	40	54	54	54	54	-	_	
7.s	2.3	-		-	_	-	-	-	48.7	48.7	5i
Laminac		4.8	4.8	-	-	_	-	_	_	_	•
Nitroceilulose, 12.6%		-	-		2	2	2.6	2	-	4	
Resction Energie	• •										
∖ H _R ' °		1912	1608	1877	~	-	-	1684	510	-	
AH _C 4		2908	3745	2010	-	-	-	3756	1203	1.66	•

as1 microns

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b180 microns

CHeat of reaction, I atm helium

dHeat of combustion, 30 atm oxygen

TABLE 2

Emission and Burning Characteristics of Flore Compositions At Verlous Simulated Altitudes*

	Approximate	Altitude	No.	Burning	Burning	Pook Energy	Total Energy	Total EH	i e ion cy
Composition	Dimensions, in.	10 ² feet	items	Time, sec	Rate sec/in.	10° watts	10° joules	faulca/e	joulos/cc
FY-719	1.35×1.5	S L°	2	5.3	3.5	2.3	12	220	337
		60	1	16.0	10.7	0.4	6	110	167
		100	3	28.0	19.9	-	-	-	-
FY-698	1.35 × 1.5	SL	1	8.5	5.6	4.0	35	630	980
		60	2	22.8	15.2	0.4	8	150	240
		100	1	50.5	33.6	0.1	6	110	176
FY-893	1.3 × 3.0	SL	1	9.8	3.4	3.4	. 39	300	630
		60	1	37.0	9.0	0.2	6	35	69
FW-168	1.35 × 1.5	SL	1	7.6	4.9	19.0	144	2440	3920
		60	2	16.5	10.5	6.0	99	1670	2710
		100	1	5 6.0	36.1	.9	33	560	902
S I -119	1.3 × 6.2	SL	1	5.0	0.8	8.0	40	90	300
	1.3×6.5	60	ţ	4.8	0.7	22.0	126	280	900
	1.3 × 5.3	100	<i>:</i>	4.0	6.8	20.0	91	210	790

⁸All flares berned from one end in cigarette fashion. To achieve this, Compositions FY-719, FY-698, and FW-168 were made as caseless flares and the curved sides were costed with Laminac resin (80/20, Nov. 4116 - 4134). FY-893 and SI-119 were pressed into this-walled (½ inch) war-created paper cases.

.

^bComposition formulas are given in Table 1 (p 7).

CSea level

TABLE 3

A Comparison of the FW-168 and SI-119 Compositions for the First 12 Seconds of Burning

				Peak	Peak Enorgy		Efficiency	
Composition	Altitude 10° feet		Approximate Longth, in.	10° wetts in 1st 12 sec	Watte/sq in. Burning Surface	10° isules in 1st 12 sec	joules/g in lat 12 sec	joules/cc in 1st 12 sec
F V -168	S L b	7.6	1.5	19.0	13 3	144	2,440	3,920
	60	16.5	1.5	6.0	4.2	72	1,220	1,920
	100	56.0	1.5	0.9	.63	7	120	190
SI-119	SL	5.0	6.2	8.0	6.	40	90	300
	60	4.8	6.5	22.0	16.6	126	280	900
	100	4.0	5.3	20.0	15.1	91	210	790

*Composition formulas:

FW-168	Teflon	46 pts
	Mg	54 pen
	NC	2 pts
SI-119	MoO ₁	31.3 pts
	Cr,O,	20.0 pts
	Ze	48.7 pcs

bses level

3

leniter Composition	Flore Compesition	No. of Itoms Jested	Approximate Dimensions, in.	Altitudo. 10º ft	; Romerk s
SI-122	FW-155	4	1.35 × 1.6	Si c	All ignited
SI-121	F#-155	3	н	60	All ignited
51-122	FW-155	13	ч	80	FW-155 ignited in 6 cases d
51-122	FW-155	12	h	100	FW-155 ignited in 6 cases
SI-122	FW-156	1	1 35 × 1.75	SL	All ignited f
SI-122	FW-156	3	•	60	All ignited
SI-12>	FW-156	ī		80	All ignited
SI-122	FW-156	1	ч	100	All ignited ^f
SI-122	FW-160	2	2.5 × 4.3	60	An ignited
SI-131	FW-155	2	1.55 × 1.6	SL	A'l ignited
SI-131	F-#-155	2	•	60	All ignited
SI-131	FW-155	;	*	80	ll ignited
SI-131	FW-155	15	N	100	FV-155 ignited in 8 cases

^aThe igniter compositions were applied to the circular aurface of the raseless flares. The flat ends were not coated.

DComposition formulas are given in Table 1 (v 7).

C Sca level

d All flares in this group had hard glased surfaces.

^{*}All flaces which failed to ignite had and, glased surface. All flaces (5) having rough, porous surface did ignite. One flace having hard, glased surface also ignited.

These I' was had mugh, porous surfaces.

SAll flages which failed or i also had hard, glanne surfa e.

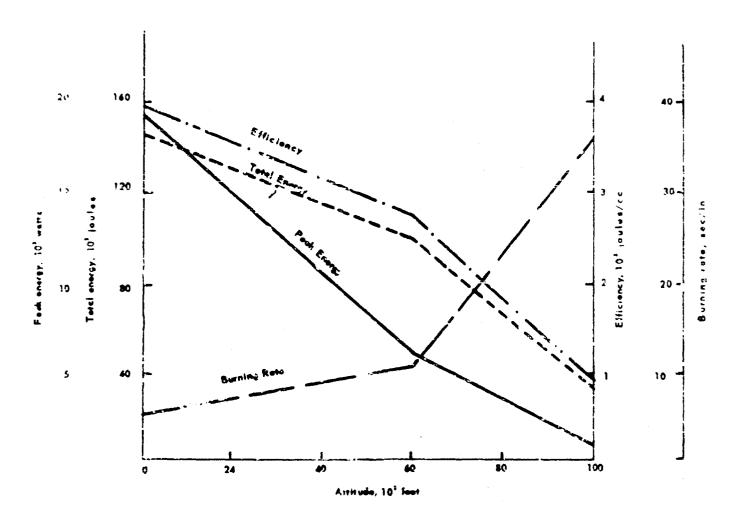


Fig 1 Effect of Altitude on the Infrared Emission and Burning Race of FW-168 (toffon/magnesium/nitrocellulcae)

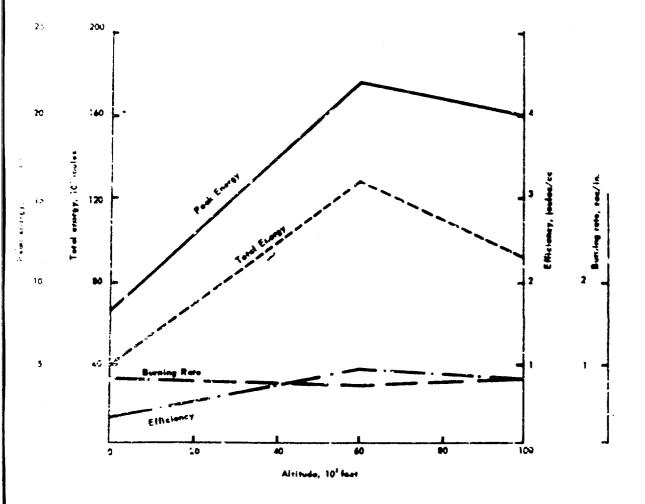


Fig 2 Effect of Altitude on the Infrared Emission and Burning Rate of SI-119

(MoO₁/Cr₂O₄/2x)

Fig 3 Pyrotechnic High-Altitude Test Chambers for Dynamic Testing at Altitudes up to 150,000 Feet (Tank volume, 8000 cubic feet)

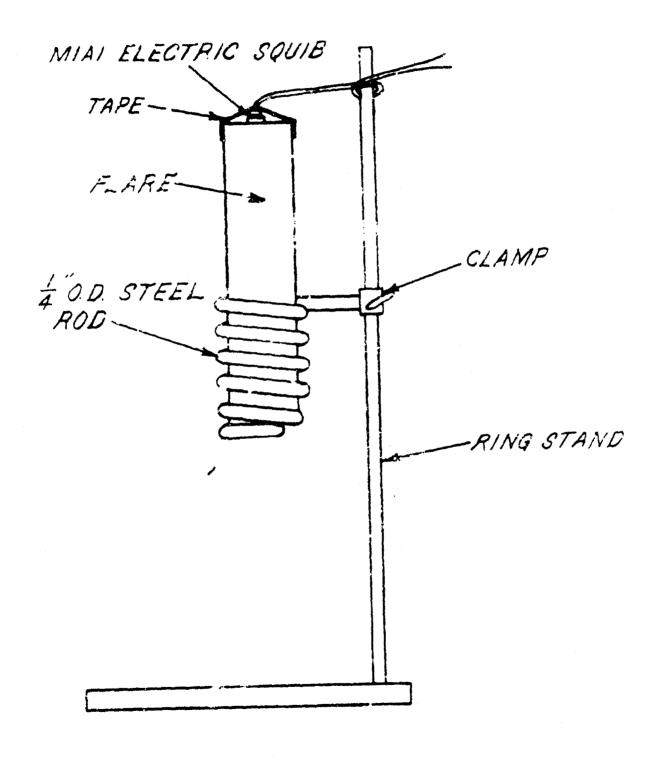


Fig 4 Static Burning Test Fixture for Caseless Flare

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